Analysis of the applicability of a common rail pump for an aircraft engine

The paper presents an analysis of the possibility of using a common rail pump to supply an aircraft compression-ignition engine. It is an engine with a two-stroke cycle, three cylinders, opposing pistons and 100 kW power. Its each combustion chamber is supply by one or two injectors controlled by electromagnetic valves. In order to assess the possibility of using a common rail pump, four high-pressure pumps were tested on a bench. They are piston pumps differing in the number and geometry of their pumping sections. The analysis included the pumping output, the torque on the pump drive shaft and the power needed to drive the pump. The weight and overall dimensions of the pump were also considered, including the arrangement of the pumping sections and the way the drive is transmitted. The research allowed to optimize the engine power supply system depending on fuel demand and the way the pump is mounted on the engine.

Key words: diesel engine, two-stroke, opposing piston, fuel pump, common rail

1. Introduction

Demanding requirements to reduce exhaust emissions and fuel consumption are increasingly affecting internal combustion engines for non-motorization applications. At the same time, the aim is to increase the power factor. As a result, intensive research work is underway to develop a diesel-powered engine for the propulsion of aircraft. Due to a number of advantages, such as: lack of the head (less heat loss) and the camshaft system, opposing piston movement conducive to engine balancing, the piston opposed diesel engine operating in a two-stroke cycle was developed and modernized [1–3]. Obviously, such a construction also has drawbacks. The main one is the need to use a gear that connects two crankshafts or a complex crank system with one shaft. The development of these engines so far has been limited by regulations on meeting the emission standards for particulate matter and hydrocarbons [4]. With the development of electronics and materials engineering, these engines are now experiencing a renaissance. The optimistic research results [5–7, 24] intensified the work on the development of efficient two-stroke diesel engines [8, 9, 25]. One of the ways to eliminate unfavorable exhaust emissions while reducing fuel consumption is to use one of the most modern fuel supply systems. The development in this field has allowed the development of a high-pressure, common rail diesel fuel supply system for diesel engines which is widely used in the automotive industry. Compared to previous solutions, it is more flexible in terms of injection, high fuel pressure generation and control and reduced energy demand [10–12].

The literature reports on some research into the injection process and related fuel dynamics [13–15, 23], maintenance of a given fuel pressure in the tank [16–18] or an injection control strategy [19–21].

An equally important part of the fuel system is the high-pressure pump. In the case of aircraft engines, no new pumps are built, but the proven solutions available on the automotive market are used. The market offer are high pressure pumps, manufactured mainly by Bosch, Denso, Delphi, Siemens. Applying this solution in aviation, the safety aspects of aircraft maintenance and operation need to be remembered. On the one hand, the smallest and lightest components must be used, but on the other hand, it must ensure safe and reliable operation of the engine. This gives rise to the dilemma of which pump to choose: single-section, small, light and adjusted to the demand, or a large and heavy multi-sectional pump with oversized efficiency. The required pumping output will, of course, depend on the configuration of the fuel system, namely the type and number of injectors.

The paper presents the results of the tests of high pressure pumps for a two-stroke compression-ignition engine with opposing pistons. The engine has three cylinders, power about 100 kW at shaft speed of 3800 rpm. Its each combustion chamber be supply by one or two injectors controlled by solenoid valves. The research allowed us to optimize the fuel supply system in terms of both efficiency and weight as well as the manner it is installed on the engine and power is transmitted.

2. Experimental apparatus and procedure

2.1. Test bench

The research was carried out on a test stand for testing common rail system elements. This is the STP/W3 test stand (Fig. 1) which is equipped with additional control and

Fig. 1. Common rail system component test bench
measurement systems. Such a test bench enables tests of injectors, high pressure pumps, dosing valves, fuel pressure control valves and pressure sensors. Universal mounting holders allow you to test varied pumps and injectors. High pressure pump is driven with a three-phase 4 kW electric motor controlled by an inverter. The temperature stabilization system enables long-term measurements in constant thermal conditions of the fuel. The diagram of the measuring system is shown in Fig. 2.

In order to determine the characteristics of high-pressure pumps, the test stand was equipped with control and measurement systems. Control of the common rail system was performed using National Instruments data acquisition cards and the LabView software. This special control and measurement system includes:

- **control of actuators:**
  - fuel dispenser in the high-pressure pump,
  - fuel pressure regulator in the fuel rail,
- **measurements:**
  - pump flow rate,
  - fuel pressure in the rail,
  - torque on the pump shaft,
  - pump shaft speed,
- **recording and data acquisition system.**

To control the actuators, NI-9758 card is used. It allows to control PWM (0–100%) up to four solenoid valves at a maximum current load of 1.5 A (continuous operation), a frequency operation of 2 Hz–10 kHz, a resolution of 500 ns, short-circuit and open circuit detection, supplied from an external source with a voltage of 7–32 V [26].

The control system is based on the CompactRIO-9024 controller. It is a real-time control and data logging device based on a 800 MHz clocked CPU for use in research systems. The platform has a built-in network interface with a data transfer rate of up to 1000 Mbps which enables remote communication and a retrieval of data logged by the device. It has a built-in 4 GB memory and 512 MB DRAM [26].

High pressure pumps are testes with a special program developed in the National Instruments LabView software. Due to hardware requirements, the program had to consist of two modules. The first one was responsible for synchronizing and communicating the actuator cards with the real time controller using DMA channels. The second module, FPGA Xilinx Virtex-5 LX110, is built in a chassis cRIO-9118. It has programmable logic blocks called CLB, I/O type blocks and programmable internal links. The possibilities of programming a particular block and the connections between the blocks are stored in the internal RAM memory.

The voltage measurement signals were recorded with the NI 9205 card. It is a sixteen-channel measuring card with a resolution of 16 bits and a maximum sampling rate of 250 kS/s. The signal level is ±10 VDC and the absolute accuracy for the range is ±10 VDC at a full scale of 6.230 μV [26].

High-pressure pumps were measured with rotor fuel flow meters: DHGF-2 (Fig. 3) and DHGF-4 depending on maximum pump flow rate. Basic technical data of the flow meters is presented in Table 1.

![Fig. 2. Diagram of the test bench for testing Common Rail system components [28]](image)

![Fig. 3. Flow meter DHGF-2 from Meister [29]](image)

<table>
<thead>
<tr>
<th>Parameter</th>
<th>Unit</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>Operating pressure, max.</td>
<td>MPa</td>
<td>1.0</td>
</tr>
<tr>
<td>Burst pressure (22°C)</td>
<td>MPa</td>
<td>&gt; 3.0</td>
</tr>
<tr>
<td>Operating temperature</td>
<td>°C</td>
<td>0–80</td>
</tr>
<tr>
<td>Measuring accuracy</td>
<td>%</td>
<td>≤±2 of the measured value</td>
</tr>
<tr>
<td>Repeatability</td>
<td>%</td>
<td>≤±0.8 of the measured value</td>
</tr>
<tr>
<td>Viscosity range</td>
<td>cSt</td>
<td>1–10</td>
</tr>
<tr>
<td>Sensing principle</td>
<td></td>
<td>Hall effect, contact-free measuring technique</td>
</tr>
<tr>
<td>Process connection</td>
<td></td>
<td>Threaded G 1/4 or 5/8 UNF</td>
</tr>
<tr>
<td>Power supply</td>
<td>VDC</td>
<td>4.5–24</td>
</tr>
<tr>
<td>Output signal</td>
<td></td>
<td>Square wave push-pull output stage</td>
</tr>
<tr>
<td>Max. output current (at 24 V)</td>
<td>mA</td>
<td>11</td>
</tr>
<tr>
<td>Measuring range</td>
<td></td>
<td>DHGF-2: 1.5–100 dm³/h</td>
</tr>
<tr>
<td></td>
<td></td>
<td>DHGF-4: 6–250 dm³/h</td>
</tr>
</tbody>
</table>

To determine the power required to drive the pump, a measuring system was built. Accordingly, the STPiW3 test stand was modernized and equipped with a torque and pump shaft speed measurement system (Fig. 4). The tested pump is mounted on a mounting plate mounted on a frame. The frame and the bearing housings are pin-connected. The axis of rotation of the frame is collinear with the axis of rotation of the pump drive shaft, which enables the transfer of torque from the pump casing to the frame through the plate. This is a universal measuring system capable of testing different pump designs. The pendulum-mounted frame is connected to the station structure by means of a ZEMIC H3-C3-75 kg strain gauge. Torque was measured by measuring the force acting on the spot where the strain gauge is mounted, located 100 mm from the axis of rotation and perpendicularly to the axis of rotation of the pump shaft. Basic technical data of the strain gauge is presented in Table 2.
2.2. Scope of the tests

The tests were carried out under the following conditions:
- temperature of the calibration fluid – 40°C,
- preliminary pump supply pressure – 0.4 MPa,
- high-pressure pump shaft speed – 500–2500 rpm,
- pumping pressure (in the rail) – 0–140 MPa.

The preset pressure was maintained by means of a solenoid valve and was the result of the PID controller. During the tests, each measurement point was defined by pump’s shaft rotation speed and the pressure in the rail. The values of pump volumetric flow rate and average torque on the camshaft were recorded four times in 10-second intervals. Additionally, torque, fuel pressure and pump shaft speed with a frequency of 2 kHz were recorded. The average torque value enabled us to specify the power required to drive the pump and then the specific energy demand to obtain the given value of flow rate.

2.3. Subject of the research

The pumps used in the research are widely applied Bosch pumps of different generations and the Denso pump:
1. Bosch CP1_H3,
2. Bosch CP4_1,
3. Bosch CP4_2,
4. Denso HP3.

These are one, two and three section positive displacement pumps with a single or double cam roller. In CP1, the sections are arranged in the body of its circumference every 120°, in CP4_2, they are shifted to each other by 90° and in HP3 by 180°. Compared to the previous solutions (CP1), the CP4 series has an optimized design by reducing the number of components and using an aluminum pump casing. High fuel pressure is generated in the pumping section and then forwarded directly through high pressure lines to the rail. There are no channels in the body for high-pressure fuel flow as it is in the CP1 pump. The same is true for HP3, where the pumping sections are connected by an external steel pipe. The pump body is made of aluminum alloys. Unlike CP pumps, the HP3 pump has a trochoid type pre-pump. The basic technical data of the tested pumps is presented in Table 3.

All pumps are equipped with a dosing valve that allows the pump output to be controlled independently of shaft speed. The valve is in the pump casing. The pumps have a flanged mounting and a possibility to place a gear wheel on the pump shaft. This enables us to install the pump in the engine block and drive it by a gear or belt drive.

3. Results and discussion

Table 4 summarizes the required volumetric flow rates of a high-pressure pump. The calculations were made for the engine operation at idle and maximum load conditions and for the option of supplying with three and six injectors (one or two per combustion chamber). The required volumetric flow rate includes injection volume, fuel from leaks and the process of injector control and 25% of the maximum fuel consumption.

Table 4. Required high-pressure pump volumetric flow rates

<table>
<thead>
<tr>
<th>Type of operation</th>
<th>Injection pressure</th>
<th>No. of injectors</th>
<th>Volumetric flow rate</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>[MPa]</td>
<td>[6]</td>
<td>[dm³/h]</td>
</tr>
<tr>
<td>idle</td>
<td>30</td>
<td>3</td>
<td>3.45</td>
</tr>
<tr>
<td>max. load</td>
<td>140</td>
<td>6</td>
<td>54.37</td>
</tr>
</tbody>
</table>

Figure 5 presents the analysis of the required volumetric flow rate with the volumetric flow rate of the tested pumps. The horizontal blue lines (idle) and red lines (max. load) illustrate the required flow rate demanded from by a high pressure pump. When the engine is supplied by three injectors, each of the tested pumps has the required volumetric flow rate, but if supplied by six injectors, the CP4.1 pump does not provide such a volumetric flow rate at maximum load.
The HP3 pump consumes the most power (Fig. 6). At a pumping pressure of 30 MPa, it is from about 0.24 kW (500 rpm) to about 1.5 kW (2500 rpm). However, at a pumping pressure of 140 MPa, it is about 1.0 and 5.0 kW, respectively. The lowest power is consumed by pump CP4.1, and then CP1_H3 and CP4.2. It is respectively about 35 and 76% of the power of pump HP3. The power demand increases linearly as the pump shaft speed increases. Obviously, the differences in the power demand to drive the pump are due to the different geometries of the pumping sections and associated the volumetric flow rates. Therefore, the further part of the paper analyzes the specific flow rates of the pumps.

Figure 7 compares the specific pumping rate for pumps operating at a pumping pressure of 30 MPa and 140 MPa. At a pumping pressure of 30 MPa, the largest specific flow rate of approx. 128 dm³/kWh is registered for the CP1.H3 pump, followed by approx. 118 dm³/kWh for the CP4.1 and HP3 pumps, and finally 112 dm³/kWh for the CP4.2 pump. However, the CP4.2 pump keeps the specific flow rate at a similar level throughout the shaft speed range. The other pumps, on the other hand, achieve the maximum specific flow rate in the speed range of 500–1500 rpm, while with increasing speed, the flow rate decreases by about 20%. With pumps operating at maximum pressure, a similar tendency to change the specific flow rate can be observed. However, maximum values of about 30 dm³/kWh for CP1, CP4.1 and HP3 can be observed for 1000–1500 rpm. For higher and lower speeds, the flow rate decreases. For CP4.2, the maximum flow rate is reached at maximum speed. The difference between the max and min values is about 7% for CP1 and CP4.2 and about 15% for CP4.1 and HP3.

Figure 8 shows the torque on the shafts of the tested pumps. This is a sample for a pumping pressure of 30 MPa and 2500 rpm. This figures enabled us to specify the values of the amplitude as well as the minimum and maximum torque on the pump shaft. The calculations were made for the pumping pressures of 30 and 140 MPa.

The lowest torque amplitude value was registered for the CP1_H3 pump. The torque amplitude analysis (Fig. 7)
shows that the lowest value is obtained for the CP1 pump at both low and high pressures and at low and high speeds (minimum approx. 3 Nm, maximum approx. 28 Nm). A comparable maximum value was also recorded for CP4.2. The highest torque amplitude values can be observed for CP4.1 and HP3. For a pumping pressure of 30 MPa, it is about 39 Nm, and for a pumping pressure of 140 MPa about 59 Nm.

Fig. 3. Torque amplitude on the pump shaft at a pumping pressure of 30 and 140 MPa and pump shaft speeds of 500 and 2500 rpm

Figure 10 and Fig. 11 show the minimum and maximum torque values in relation to the pumping pressure and the pump shaft speed. The most regular torque, mostly of positive or slightly negative (approx. –3 and –7 Nm) values, over the entire operating range is for the CP1 and CP4.2 pumps. The second group is the CP4.1 and HP3 pumps. The torque takes both positive and negative values. The CP4.1 pump has higher negative values of approx. –25 Nm at high speeds. The highest positive torque of the HP3 pump (about 53 Nm) at 140 MPa and 2500 rpm.

Fig. 10. Minimum and maximum torque at 30 MPa pumping pressure and 500 and 2500 rpm shaft speeds

4. Summary

The tests show that to power an engine of a power of about 100 kW and with three injectors (one per combustion chamber), all three pumps have the required flow rate, both at idle and maximum load conditions. If six injectors (two per combustion chamber) are used, the CP4.1 pump does not provide the required capacity at maximum load.

Analyzing the specific flow rate, it can be stated that there are smaller differences between the tested pumps at the maximum pumping pressure. It is also observed that the CP4.2 pump has an almost constant the specific flow rate in the entire speed range. The efficiencies of the other pumps are highest in the range of 500–1500 rpm.

The smallest torque amplitude is recorded for the CP1 pump. There is a positive torque almost over its entire operating range, which ensures that the drive system is loaded with one-way forces. Another advantage is the number of pumping sections, allowing the synchronization of pumping and fuel injection. However, its large mass and no individual high-pressure connections disqualifies it from mounting in the aircraft engine.

The best option is the CP4.1 pump due to its small mass. However, it has one pumping section, which cannot guarantee the required safe operation of aircraft. With this in mind, a pump with a minimum of two pumping sections and pressure individually discharged into the fuel rail/rails can be applied, which makes the system more reliable. Pumps CP4.2 and HP3 satisfy these requirements. The required flow rate is achieved in the speed range of 1500–2000 rpm. This allows the use of a 1:2 pump drive ratio in relation to the engine crankshaft, reduction of rotation and operation with the highest efficiency (specific flow rate). The pumps are flanged and can be fitted with a toothed wheel or a belt. Accordingly, a toothed or belt transmission can be used and the pump can be more flexibly mounted on the engine with other engine equipment. The advantage of HP3 is the opposite position of its pumping sections and a built-in pre-feed pump. The disadvantage of the pump is that the flow rate decrease with increasing speed (especially at 30 MPa) and higher torque amplitude.

Acknowledgements

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Bibliography

[27] https://www.zemiceurope.com
[29] https://meister-flow.de

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